Research Article

Design of Experimentation of Engine Head Bolt by Finite Element Analysis

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Abstract

This paper represents the design of the engine head bolt by Finite element analysis (FEA). Response surface modeling and analysis of compressive ignited engine head bolt carried out by varying different parameters. Design parameters such as thread radius, total length, thread angle, force, and temperature at various ranges. Simple equation is calculated which gives values of total deformation of compressive ignited engine head bolt by carrying out a regression analysis. It shows that force, thread angle and material property are significant parameters which affect total deformation because their P value near to 0.92. Also, it shows the design parameter effect of deformation of the engine head bolt has been obtained.

Keywords: Engine head bolt, Response surface modeling, deformation, composite material etc.

1. Introduction

Bolted joints are one of the most common elements in construction and machine design. They consist of fasteners that capture and join other parts, and are secured with the mating of screw threads. The compression Ignited engine head bolt having most common mode of failure is due to overloading. Operating forces of the application produce loads that exceed the clamp load, causing the joint to loosen over time or fail catastrophically.

Over torque might cause failure by damaging the threads and deforming the fastener, though this can happen over a very long time. Under torque can cause failures by allowing a joint to come loose and it may also allow the joint to flex and thus fail under fatigue. Brinelling may occur with poor quality washers, leading to a loss of clamp load and subsequent failure of the joint. Other modes of failure include corrosion, embedment, exceeding and the shear stress limit.

2. Simulation

Finite element analysis of engine head bolt

Analysis has been carried out in ANSYS work bench. Constrained geometry of the engine head bolt is as shown in figure 1.1.



Fig.2.1 Geometry of the bolt in ANSYS

Fig.1.2 shows that the geometry of bolt is meshed by using solid element. Total number of nodes 49208 and elements 27491 are generated after meshing. One end of the bolt is fixed in all direction while load is applied in another end.



Fig.2.2 Meshed model of engine head bolt

Table No.1.1 shows that the Aluminum Matrix Sic Fiber (17.5%) composite metal matrix material properties is selected from the engine specification requirement as per follows

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| Sr.No | Material Property | Values | |
|-------|--|----------|--|
| 1 | Density (kg/m3) | 2800 | |
| 2 | Young's modulus (Pa) | 1.00e11 | |
| 3 | Poisson's ratio | 0.22 | |
| 4 | Bulk modulus (Pa) | 5.95e10 | |
| 5 | Shear modulus (Pa) | 4.09e10 | |
| 6 | Ultimate tensile strength (Pa) | 4.616e+8 | |
| 7 | Yield tensile strength (Pa) | 4.065e+8 | |
| 8 | Yield comp. strength (Pa) | 4.065e+8 | |
| 9 | Coefficient of thermal expansion (1/°C×10-6) | 14 | |

Table No.2.1 Material properties of Bolt for analysis

Table No.2.2 Corresponding parameters according to RSM (Response Surface Modelling)

| Sr.No. | P1 | P2 | P3 | P7 | P8 | P5 | P6 |
|--------|----------|----------|----------|----------|----------|----------|----------|
| 1 | 149.1212 | 54.11802 | 4.119818 | 264.9443 | 29592.73 | 0.595255 | 1139.934 |
| 2 | 157.7194 | 59.51168 | 4.021793 | 258.3032 | 29018.66 | 0.61527 | 1218.365 |
| 3 | 149.6792 | 55.70533 | 4.40143 | 238.7408 | 31036.18 | 0.626759 | 976.4385 |
| 4 | 154.7907 | 58.40598 | 4.242438 | 241.4925 | 29973.01 | 0.624008 | 1117.496 |
| 5 | 131.9644 | 57.54947 | 3.912565 | 266.4022 | 27742.43 | 0.49775 | 927.8082 |
| 6 | 135.58 | 62.89415 | 4.295003 | 247.2307 | 26926.92 | 0.495813 | 957.4723 |
| 7 | 144.0676 | 60.6398 | 4.512688 | 233.9464 | 27152.41 | 0.528952 | 1049.679 |
| 8 | 158.3936 | 65.03706 | 3.929235 | 251.1106 | 28226.89 | 0.60087 | 911.4311 |
| 9 | 140.283 | 62.03025 | 3.821967 | 225.5161 | 31166.07 | 0.592667 | 1187.417 |
| 10 | 142.6046 | 63.94756 | 4.479751 | 270.3668 | 31984.25 | 0.617064 | 1446.358 |
| 11 | 148.2485 | 60.56134 | 4.376401 | 254.8228 | 31927.2 | 0.639169 | 1041.985 |
| 12 | 136.4377 | 55.64097 | 4.562373 | 232.5476 | 29804.66 | 0.551768 | 1315.098 |
| 13 | 132.5122 | 56.64209 | 3.99268 | 272.5 | 27826.4 | 0.500743 | 1028.232 |
| 14 | 155.94 | 64.16241 | 4.510396 | 246.0863 | 26604.47 | 0.558297 | 904.2802 |
| 15 | 152.6734 | 63.16849 | 3.807754 | 244.4476 | 31545.33 | 0.648406 | 1036.967 |
| 16 | 160.4687 | 54.78512 | 4.186982 | 264.9663 | 30191.04 | 0.65049 | 1067.669 |
| 17 | 144.2155 | 59.37968 | 4.023478 | 266.7081 | 28470.99 | 0.554806 | 1268.877 |
| 18 | 151.1215 | 62.15171 | 4.105871 | 236.5765 | 27144.43 | 0.55288 | 1000.732 |
| 19 | 141.665 | 58.20628 | 3.901381 | 227.8357 | 29302.87 | 0.562227 | 1270.166 |
| 20 | 138.2017 | 65.59567 | 4.33174 | 256.2937 | 30971.42 | 0.580462 | 1179.807 |
| 21 | 148.7863 | 65.97119 | 3.767004 | 247.0326 | 29834.79 | 0.599324 | 1264.182 |
| 22 | 134.2455 | 54.17126 | 4.211562 | 257.8139 | 30891.93 | 0.56273 | 1089.27 |
| 23 | 153.2576 | 59.60686 | 4.315126 | 232.3447 | 31433.7 | 0.648845 | 1101.996 |
| 24 | 156.0447 | 55.59037 | 4.159218 | 273.7408 | 30427.57 | 0.638643 | 1273.025 |
| 25 | 138.3603 | 61.0834 | 4.557762 | 251.3462 | 26563.29 | 0.498904 | 1057.34 |
| 26 | 137.3967 | 62.17072 | 3.902451 | 268.0481 | 28632.67 | 0.533508 | 1158.763 |
| 27 | 144.7068 | 57.4149 | 3.956585 | 226.459 | 27146.66 | 0.53093 | 1127.238 |
| 28 | 151.535 | 58.07879 | 4.011851 | 236.0168 | 28962.86 | 0.591765 | 1121.448 |
| 29 | 160.9636 | 62.85359 | 4.362881 | 262.2884 | 27836.88 | 0.602558 | 894.6289 |
| 30 | 140.7004 | 64.10964 | 4.453999 | 243.2682 | 31720.51 | 0.604649 | 1177.957 |
| 31 | 155.5096 | 58.23349 | 4.306086 | 256.9845 | 26691.79 | 0.55853 | 1016.306 |
| 32 | 152.5525 | 56.24437 | 3.776942 | 253.1633 | 29017.78 | 0.596172 | 1289.63 |
| 33 | 136.5402 | 63.52509 | 4.053729 | 228.1196 | 30319.4 | 0.561838 | 1248.532 |
| 34 | 132.3653 | 55.00231 | 4.192097 | 241.1152 | 27780.65 | 0.499765 | 1117.542 |
| 35 | 158.7122 | 57.02604 | 4.397346 | 248.837 | 31825.21 | 0.678632 | 1050.873 |
| 36 | 151.967 | 65.11317 | 3.952788 | 230.6938 | 28205.59 | 0.578037 | 974.2129 |
| 37 | 148.6488 | 58.92983 | 4.544701 | 235.8949 | 29655.41 | 0.59483 | 1056.409 |
| 38 | 143.1115 | 63.64291 | 4.447756 | 272.885 | 27131.96 | 0.525611 | 947.6484 |
| 39 | 144.7667 | 61.38987 | 3.897125 | 266.8779 | 30869.78 | 0.604106 | 1360.887 |
| 40 | 140.4207 | 60.27452 | 4.124818 | 261.7584 | 31117.61 | 0.592494 | 1389.237 |
| 41 | 141.0624 | 61.22917 | 3.891098 | 231.2955 | 30192 | 0.576616 | 1211.342 |
| 42 | 134.4648 | 58.17288 | 4.291296 | 238.6028 | 32016.46 | 0.584525 | 1310.076 |
| 43 | 152.8452 | 62.64547 | 4.1//13/ | 244.0125 | 28145.93 | 0.579313 | 901.8488 |
| 44 | 152.237 | 57.2219 | 4.48534 | 228.3353 | 26434.84 | 0.542197 | ///.429 |
| 45 | 140.5343 | 60.83981 | 4.136523 | 249.3524 | 28390.82 | 0.540583 | 1289.924 |
| 40 | 145.214/ | 59.81205 | 3.994563 | 270.9988 | 2/08/.5/ | 0.53154/ | 1048.794 |
| 47 | 158.1035 | 56.13542 | 3.812/48 | 254.613/ | 30980.54 | 0.658148 | 1253./61 |
| 48 | 158.4853 | 65.53935 | 4.421408 | 258.4833 | 31290.14 | 0.666294 | 1025.496 |
| 49 | 147.0521 | 54.02/52 | 4.593499 | 265.0111 | 29650.55 | 0.588822 | 1066.481 |
| 50 | 137.6982 | 64.27583 | 4.010399 | 262.8771 | 29100.46 | 0.543437 | 925.1188 |

2140| International Journal of Current Engineering and Technology, Vol.6, No.6 (Dec 2016)

Design & Analysis of Heat Sink High-Pressure Die Casting Component

| 51 | 153.9449 | 62.06073 | 3.986306 | 273.7326 | 29549.28 | 0.612944 | 904.9875 |
|----|----------|----------|----------|----------|----------|----------|----------|
| 52 | 156.9775 | 54.55229 | 4.41073 | 261.9411 | 27413.1 | 0.57879 | 942.6765 |
| 53 | 143.9357 | 59.99305 | 4.589466 | 243.6204 | 28573.07 | 0.555919 | 986.022 |
| 54 | 140.8886 | 63.05483 | 4.083594 | 237.3483 | 26540.03 | 0.506603 | 1042.806 |
| 55 | 152.2359 | 56.70674 | 4.12285 | 227.0305 | 31724.77 | 0.650486 | 1024.889 |
| 56 | 135.4342 | 60.94513 | 3.911477 | 248.5502 | 27968.89 | 0.514378 | 1108.173 |
| 57 | 133.0315 | 56.00557 | 4.281729 | 254.9469 | 30150.29 | 0.545085 | 1001.212 |
| 58 | 147.1384 | 58.48783 | 3.784858 | 256.8256 | 30538.15 | 0.60661 | 1178.369 |
| 59 | 138.3457 | 64.31881 | 4.461243 | 267.2525 | 31542 | 0.591987 | 1055.114 |
| 60 | 160.639 | 65.66412 | 4.239269 | 234.6363 | 29245.7 | 0.631282 | 1197.184 |
| 61 | 132.2402 | 65.04582 | 4.247 | 257.2642 | 27405.56 | 0.492549 | 1064.493 |
| 62 | 136.0951 | 58.17916 | 4.464921 | 262.7362 | 28989.04 | 0.535576 | 1088.398 |
| 63 | 153.8936 | 62.98892 | 4.575373 | 231.7649 | 30158.86 | 0.625049 | 1339.586 |
| 64 | 141.767 | 57.13644 | 3.922864 | 236.551 | 29908.52 | 0.574351 | 1128.632 |
| 65 | 145.7687 | 55.64686 | 4.036151 | 248.287 | 26928.71 | 0.530205 | 1044.549 |
| 66 | 159.2674 | 61.1415 | 4.275542 | 271.4136 | 28622.31 | 0.613081 | 914.0388 |
| 67 | 151.3902 | 59.37288 | 3.807137 | 266.8518 | 31195.65 | 0.636898 | 1026.614 |
| 68 | 147.0665 | 54.41851 | 4.356002 | 244.605 | 30704.58 | 0.609782 | 955.5727 |
| 69 | 156.6605 | 61.63907 | 3.949441 | 227.2955 | 28004.94 | 0.590302 | 1004.931 |
| 70 | 138.0044 | 63.93182 | 4.100436 | 252.1667 | 31852.58 | 0.595825 | 1010.069 |
| 71 | 155.9486 | 55.23448 | 4.120167 | 240.3435 | 27447.12 | 0.575671 | 893.7658 |
| 72 | 133.4058 | 59.19917 | 3.938626 | 253.2439 | 28876.21 | 0.523154 | 918.3957 |
| 73 | 159.3734 | 60.42202 | 3.878969 | 238.0182 | 32256.31 | 0.691289 | 1176.375 |
| 74 | 140.4794 | 62.72935 | 4.270048 | 265.6901 | 31424.12 | 0.598572 | 1249.872 |
| 75 | 154.9857 | 58.05715 | 4.450188 | 273.9779 | 27912.75 | 0.582205 | 1040.942 |
| 76 | 142.1449 | 54.68441 | 4.065498 | 263.4347 | 30069.85 | 0.578714 | 1043.452 |
| 77 | 147.5122 | 62.11699 | 4.411 | 245.16 | 30526.03 | 0.608067 | 1319.429 |
| 78 | 151.4036 | 64.39214 | 4.516486 | 230.1647 | 28262.86 | 0.576879 | 1008.814 |
| 79 | 137.6653 | 57.52656 | 4.230318 | 227.2772 | 29591.87 | 0.552693 | 1256.699 |

Table No1.2 shows that the relation between the numbers of input parameters and gives deformation and stress to build a response surface modeling. A Design of Experiments, or DOE, the method required many design points should be solved. Once the required solutions are complete a response surface is fitted through the results, allowing designs to be queried where no hard solution exists. In this table, P1 – Total Length, P2 – Angle of Thread, P3 – Thread Radius, P7 – Temperature (⁰C), P8 - Force Magnitude (N), P5 - Total Deformation Maximum (mm), P6 - Equivalent Stress Maximum (MPa).

As shown in the table No.1.2. Regression analysis of engine head bolt is carried out by using M S excels 2007 for obtaining regression equation. Regression analysis of engine head bolt has been done by using MS excel and following equation of regression is obtained.

P5=-0.5383+0.003702(P1)-5.9E-06(P2)-0.00109(P3) +3.18E-06(P7) +1.98E-05(P8) (1)

3. Results and discussion

In this paper, analysis of engine head bolt by using the response surface modelling has been discussed. This analysis provides the resulting graphs of design parameters Vs total deformation. Under the loading condition, total deformation of engine head bolt is obtained. Total deformation of engine head bolt is as shown below in figure 1.3.In this case, static analysis is done by using the finite element analysis, in the figure blue color indicates the minimum deformation 0.06449 mm acting on the bolt and Red color indicates maximum deformation 0.5804 mm.



Fig.3.1. Total deformation of engine head bolt

3.1 Angle Vs Total Deformation



Fig.3.1. Angle Vs Total Deformation Maximum (mm)

Figure shows relationship between angles Vs Total deformation. In this case, when angle increases total deformation of bolt increases up to 59^o and after that it decreases when increases in angle where maximum total deformation is 0.580801 mm.

3.2 Force Magnitude (N) VS Total Deformation (mm)



Fig.3.1. Force (N) Vs Total Deformation (mm)

Figure shows relationship between Force Vs Total deformations. In this case, when force magnitude increases total deformation of bolt increases up to 32263 N and after that it decreases because stress exceeds elastic limit.

3.3 Total Length (mm) Vs Total Deformation (mm)





Figure shows relationship between Total Length Vs Total deformations. In this case, when bolt Length increases total deformation of bolt increases up to 0.6345808 mm and after that it deformation increases because increasing the length.

3.2 Force Magnitude (N) VS Total Deformation (mm)





Figure shows relationship between Thread Radius Vs Total deformations. In this case, when Thread Radius increases total deformation of bolt increases up to 0.580846 mm and after that it decreases because increasing the diameter of bolt.

Conclusions

- Finite element analysis has been carried out in engine head bolt.
- 2) It is observed that thread angle and material property are significant parameters which affect total deformation because their P value is 0.92.
- Relationship among design parameters and total deformation has been obtained.
- 4) Simple equations find out which gives value of total deformation of engine head bolt by carrying regression analysis in M S office 2007.

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